

# Power Conditioning & Distribution Units

# RELIABILITY, RESILIENCE & PERFORMANCE

AAC Clyde Space's small satellite PCDU (Power Conditioning and Distribution Unit) solutions have demonstrated impressive space-based capabilities stretching across a range of applications, gaining a market reputation for reliability, resiliency and performance. STARBUCK-MINI provides high power 28 V outputs and redundancy for power distribution as well as redundant command and control interfaces via CAN or RS422. Its innovative design combines COTS and radiation hardened components to optimizes the reliability and performance characteristics of the ITAR free system. The STARBUCK-MINI adopts a modular design approach which enables easy customization. The adaptable system utilizes flexible add-ons to meet the required number of power output interfaces for payloads on each specific mission and can be quickly adapted to customer product specifications. Integration of interfaces for deployment mechanisms, magnetic torquers and other equipment can also be accomodated through the addition of custom modules.

#### POWER

With 1500 W system power, 28 V nominal bus and battery voltage with power converters for isolated auxiliary output. High-power 28V outputs & redundancy for power distribution, as well as FPGA based control & monitoring of all switches & interfaces through redundant CAN or RS422.



PERFORMANCE

High-performance power solution for small satellite platforms. Modular design approach delivers scalability and easy tailoring of interfaces to mission requirements. With MPPT or S3R battery charge regulation for advanced power management. RELIA

Reliability and qualification levels suitable for many different mission types. COTS components with verified space performance combined with radiation hardened components. All outputs protected by Latching Current Limiters (LCLs) or Retriggerable LCLs (RLCLs). With support for redundant power supply to consumers. Designed for optimized reliability with performance fail-safes, the state-ofthe-art STARBUCK-MINI offers advanced power management with 1500W system power.

## TECHNICAL SPECIFICATIONS

General	
Design Life	5-7 years in LEO
System Power	1500 W
Primary Bus Voltage	22-34 V
Secondary Bus Voltage	5, 8, 12 or 15V with
	isolated ground
Battery Regulation	MPPT or S3R
Idle Power Consumption	10W
Operating Temperature	-30°C to +60°C
Range	
Radiation (TiD)	20 kRAD
	(quali-fied >30 kRAD,Si)
Mass	3300 g

Electrical Interfaces (baseline configuration)		
Primary Bus High Power	4 individual	
(up to 10 A) Outputs	pro-tection (LCL)	
Primary Bus Nominal	12 individual	
(up to 3A) Outputs	protec-tion (LCL or RLCL)	
Secondary Bus Outputs	12 individual	
	protec-tion (LCL or RLCL)	
Solar Array Interface	4 x 120 W triple	
	junction panel strings	
	(expandable to 12 strings)	

Dimensions	
Height	147 mm
Depth	249 mm
Width	123 mm (baseline configuratiion)

All technical specifications are suject to change without notice.

Telemetry and Control Interfaces	
TM/TC	CAN or RS422 serial
	inter-face (redundant)
	Redundant flash-based FPGA
	controllers
Telemetry	Bus voltage
	Battery current
	Internal unit temperature
	Internal unit voltages
	LCL status
	Individual LCL currents
Pulse Command Reset	RS422 levels – triggers
	system-wide power cycle.
Actuators and	Arm and fire
Thermal Knives	actua-tion (bus voltage)
Separation Detection	Triple redundancy with
from Launch Vehicle	majority voting

#### Baseline configuration (4 modules) and module functionality

Module	Function	Handling	Width
BCR/APR	Battery connection point and charge	60A battery connector.	40mm
	regulator. Array power regulator.	Umbilical connection	
		Up to 4 S/A strings at 120W max each.	
CTRL	Control unit with redundant FPGAs	Redundant CAN or RS422 interfaces	22mm
		Separation detection	
		Pulse command input	
PPD	Primary power	4 class 10 main bus LCL outputs	22mm
	distribu-tion (std. option)	10 class 3 main bus LCL outputs	
		2 class 3 main bus RLCL outputs	
		(Optional configuration 20 class 3 LCL outputs)	
SPD	Secondary power distribution	2 x 30W low voltage outputs divided on	22mm
		2 x 6 individual LCLs (up to class 2)	

# TECHNICAL SPECIFICATIONS

#### Example expanded configuration (11 modules)

Module	Function	Handling
BCR/APR	Battery connection point and charge	60A Battery connector Up to 12 S/A
	regulator. Array power regulator.	strings at 120W max each.
APR	Array power regulator.	
APR	Array power regulator.	
CTRL	Control unit with redundant FPGAs	Redundant CAN or RS422 interfaces
		Separation detection
		Pulse command input
PPD	Primary power distribution (std option)	8 class 10 LCL outputs
PPD	Primary power distribution (std option)	56 class 3 LCL outputs
PPD	Primary power distribution (20 channel option)	8 class 3 RLCL outputs
PPD	Primary power distribution (20 channel option)	
SPD	5V and 12V Secondary power distribution	2 x 30W low voltage outputs
		6 LCLs for 5V (up to class2)
		6 LCLs for 12V (up to class2)
SPD	5V and 12V Secondary power distribution	2 x 30W low voltage outputs
		6 LCLs for 5V (up to class2)
		6 LCLs for 12V (up to class2)
CUSTOM	Custom functionality add-on module	High power outputs with fuses
		Magnetorquer control with preci-
		sion current controlled outputs
		Propulsion control with H-bridges, instrumentation
		amplifiers and LCLs for valves, thrusters and heaters.
		Analog signal conditioning for thermistors.



To make an enquiry, request a quotation or learn about AAC Clyde Space's other products and services, please contact: **enquiries@aac-clyde.space** 

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